

Eroding dipoles and vorticity growth for Euler flows in \mathbb{R}^3 : axisymmetric flow without swirl

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A review of analyses based upon anti-parallel vortex structures suggests that structurally stable dipoles with eroding circulation may offer a path to the study of vorticity growth in solutions of Euler's equations in \mathbb{R}^3 . We examine here the possible formation of such a structure in axisymmetric flow without swirl, leading to maximal growth of vorticity as $t^{4/3}$. Our study suggests that the optimizing flow giving the $t^{4/3}$ growth mimics an exact solution of Euler's equations representing an eroding toroidal vortex dipole which locally conserves kinetic energy. The dipole cross-section is a perturbation of the classical Sadovskii dipole having piecewise constant vorticity, which breaks the symmetry of closed streamlines. The structure of this perturbed Sadovskii dipole is analysed asymptotically at large times, and its predicted properties are verified numerically. We also show numerically that if mirror symmetry of the dipole is not imposed but axial symmetry maintained, an instability leads to breakup into smaller vortical structures.

Key words: vortex dynamics, vortex flows, vortex instability

1. Introduction

The purpose of this paper is to construct an Euler flow involving an eroding dipole structure which achieves maximal long-time growth of vorticity as $t^{4/3}$ in axisymmetric flow without swirl. Our analysis will employ asymptotic estimates and neglect certain higher-order effects, but the results will be supported by numerical calculations. Our aim is to present a plausible if approximate physical model with a number of compelling features, which enables some explicit (if formal) analyses of vorticity growth in three dimensions.

We focus here on the local amplification of vorticity, in other words on the self-stretching of a vortex structure. This is in contrast to the stretching that results from distant interactions of vortex structures. The scaling invariance inherent in Euler flows allows such local stretching to proceed in principle to arbitrarily small scales, possibly leading to extremely rapid growth of the vorticity. This viewpoint has indeed motivated much of the research into the possibility of blow up of vorticity in finite time in three dimensions and has led almost exclusively to consideration of the

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interaction of anti-parallel vortex structures. An excellent summary of this research may be found in Gibbon (2008). We mention in particular the work of Pumir & Siggia (1987) on the interaction of anti-parallel, thin vortex tubes, research which showed that such tubes tend to pair up and begin to interact. However, it turns out this interaction cannot avoid the ultimate distortion of vortex cores. This is because, considered as line vortices, the motion brings the filaments together at a rate which is proportional to the logarithm of the product of the radius and the curvature. This product must remain large to ensure the integrity of the cores. This then implies that the distance between the filaments shrinks faster than the core size, so distortion must occur and the filament model fails; for details see Hormoz & Brenner (2012).

Explicit numerical studies of core interaction in three dimensions have again involved anti-parallel tubes, see Gibbon (2008), and also Bustamante & Kerr (2008) and Hou & Li (2008). Here rapid growth of vorticity is observed. The presence or not of a finite-time singularity has been controversial, but recent analysis of higher-order norms of vorticity indicate double-exponential growth (Kerr 2013). The problem of core interaction also occurs in the simpler problem of collision of two vortex rings, see Oshima (1978), Lim & Nickels (1992) and Riley (1998). This brings us to the main focus of the present paper, the interaction of anti-parallel vortex tubes in axisymmetric flow without swirl (AFWOS). In a subsequent paper, we shall extend the discussion to include general anti-parallel structures in three dimensions.

Our approach is to use this simpler problem to explore in detail core interaction. It is well known that in AFWOS there can be no finite-time blow up of vorticity (Majda & Bertozzi 2001). Nevertheless, this is an arena where modest amplification of vorticity can be studied in detail. We have argued that this problem leads naturally to the important role played by the local conservation of total kinetic energy (Childress 2008). This enforces a loss of volume of the vortical structure associated with growth, which can then be described as an ‘eroding’, toroidal, dipolar structure. Such erosion is sometimes also described as ‘stripping’, in which a vortex loses outer layers of vorticity, thus sharpening the vorticity profile. We show that such a structure should emerge generally from equal and opposite colliding vortex rings, and that the ultimate fate can be realized by a solution of Euler’s equations corresponding to a eroding, locally two-dimensional structure having a uniform vorticity in each of the two constituent eddies. The non-eroding counterpart is the well-known two-dimensional (2-D) Sadvskii vortex with continuous velocity (Sadvskii 1971; Pierrehumbert 1980; Saffman & Tanveer 1982).

Analysis of uniform vorticity patches have an interesting history. It has long been known that steady flows with closed streamlines should, for sufficiently small viscosity, tend to regions of constant vorticity (Prandtl 1952; Batchelor 1956). In particular, Batchelor proposed that they occur in the wake of a bluff body at high Reynolds number. Solutions of Euler’s equation illustrating such eddies were given by Childress (1966) under the condition that they be slender, a constraint that did not allow continuous velocity at the boundary of the eddy but exhibited a doubly cusped limiting case. Sadvskii studied the class quite generally and obtain the example with continuous velocity (Sadvskii 1971). A interesting example of a non-slender doubly cusped Sadvskii eddy is to the stable 2-D wake behind a bluff body at large Reynolds number (Chernyshenko 1988).

There is already clear numerical evidence for the existence of such eroding structures. Studies of interactions of anti-parallel vortex rings have suggested that vorticity tends to be shed into a sort of ‘tail’ aft of the main body of the resulting dipolar vortex, as the tubes are stretched, see e.g. Pumir & Kerr (1987), Shelley,

Meiron & Orszag (1993), Bustamante & Kerr (2008) and Grafke & Grauer (2013). Calculations of colliding rings using the techniques of contour dynamics explicitly exhibit the development of a long ‘tail’ and a ‘head’ and thus a ‘tadpole’ shape for the dipole/tail structure; see Riley (1998), Shariff, Leonard & Fersiger (2008). In Shariff, Leonard & Fersiger (2008) it was shown that the head of the tadpole is indeed very close to the shape of the Sadvskii dipole. Our claim here is that this configuration emerges generally in AFWOS under the condition that we are dealing with a toroidal dipole that is anti-symmetric about a plane dividing the two vorticity regions. Note that the physical experiment described in Lim & Nickels (1992) involves vortex stretching in an apparently axisymmetric phase, before a non-axisymmetric instability develops that ends the expansion. (A video may be seen at <http://www.youtube.com/watch?v=12ozAloKYyo>.) However, prior to the instability one observes an axisymmetric ‘membrane’ which is consistent with the shedding of a tail behind the axisymmetric dipole pair propagating radially outwards.

The paper is organized as follows. Section 2 introduces the axisymmetric geometry and gives some background of the development of the dipolar structure. In § 3 we present a simplified analysis of the eroding dipole based upon the *ad hoc* temporal scalings of scale and velocity derived from the constraint of locally constant kinetic energy. In § 4 the program outlined in § 3 is subjected to more detailed asymptotic analysis in order to compute these scalings explicitly. We shall thereby derive the property of local energy conservation directly from the dynamics of a perturbed Sadvskii vortex. In § 5 numerical simulations are described which are found to exhibit the scalings of the previous sections as well as show the evolution toward the asymptotic state. We also calculate in one instance break-up of the dipole through an instability breaking the mirror symmetry. We summarize our results and indicate their implications for more general anti-parallel vortex interactions in the discussion of § 6.

2. Preliminaries

2.1. The axisymmetric geometry

In AFWOS, we may study the growth of vorticity by expansive stretching in its simplest setting. It is known that no finite-time singularity can then be formed (Majda & Bertozzi 2001), but one may still pose an initial value problem in \mathbb{R}^3 and ask how fast vorticity can grow at large times. The problem was taken up in Childress (2008), and we now summarize the results. The ideas outlined here will be developed further in § 3.

Since the vorticity consists of rings with a common axis, maximal growth of vorticity as $t \rightarrow \infty$ can be determined by considering a symmetric anti-parallel bundle of vortex rings. Optimization under the condition of conservation of vorticity volume then bounds (for large t) the maximum of vorticity as a multiple of t^2 . This estimate can be understood as follows. Imagine a torus with a centreline \mathcal{C} of radius R and a circular cross-section of radius a . Let the angular vorticity component (the only component to be considered here) be $\pm\omega_\theta$ in the two half-discs of the cross-section, the signs such as to produce expansive stretching. By conservation of volume, $a^2R \sim 1$ in order of magnitude. Also, vortex dynamics ensures $dR/dt \sim \omega_\theta a$. Finally, conservation of vorticity flux requires $\omega_\theta a^2 \sim 1$. Thus $dR/dt \sim \sqrt{R}$ leading to the t^2 estimate. We now set $\omega_\theta = \omega$ for the axisymmetric case.

The growth as t^2 cannot be obtained by solutions of Euler’s equations since the volume-conserving optimizer does not conserve total kinetic energy E . Indeed,

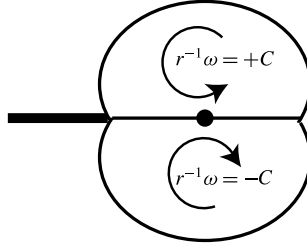


FIGURE 1. Vorticity distribution for the dipole which maximizes the velocity of the centreline ‘target’ (the centre dot), subject to constraints on volume, ω/r , and total kinetic energy. Here C denotes the maximum initial value of $|\omega/r|$.

$E \sim Ra^2(\omega a)^2 \sim R$. If conservation of energy is also imposed, and a cross-sectional scale determined again by a single length a , one must take $a \sim R^{-3/4}$ so that vorticity volume is lost. In fact we can only maintain kinetic energy approximately, with loss of volume and energy occurring through the shedding of a ‘tail’ of vorticity-laden fluid from the vortex pair, of thickness $H \sim R^{-5/2}$.

To see this, suppose then that we seek a structure with $a \sim R^{-p}$, $p > 0$, which extrudes a tail in the form of a sheet of thickness $H \sim R^{-q}$. The rate of change of dipole volume is of order $d(a^2R)/dt \sim R^{-2p}\dot{R}$ and the flux of volume into the tail is (since vorticity is proportional to R) equal to $HR\dot{R} \sim R^{1-q}\dot{R}$. Conservation of volume requires that

$$R^{-2p}\dot{R} \sim R^{1-q}\dot{R}, \quad (2.1)$$

so that $q = 2p + 1$. For the kinetic energy, considered relative to the fluid at infinity, we first compute the flux of energy into the tail. The velocity in the tail is of order $\omega H \sim R^{1-q}$ and therefore, as the circular band of height H and radius R expands at a rate \dot{R} , energy is created in the tail at a rate

$$\frac{dE_{tail}}{dt} \sim R\dot{R}(\omega R^{-q})^2 R^{-q} \sim R^{3-3q}\dot{R}. \quad (2.2)$$

If this must equal the energy decrease in the ‘head’ of the structure, estimated as

$$\frac{dE_{head}}{dt} \sim \frac{d}{dt} [Ra^2(\omega a)^2] \sim \frac{d}{dt} R^{3-4p} \sim (3-4p)\dot{R}R^{2-4p}, \quad (2.3)$$

and also $p \neq 3/4$, then $R^{2p+2} \sim 1$, which is impossible. The only recourse is to set $3-4p=0$ to make $E_{head} \sim 1$. Thus $(p, q) = (3/4, 5/2)$ and $dR/dt \sim \omega R^{-3/4} \sim R^{1/4}$, yielding a maximum growth for large t as $R \sim t^{4/3}$. Kinetic energy is lost to the tail at a rate $R^{1-7p} = R^{-17/4}$ from (2.2) and so is extremely small at large R , consistent with $E_{head} \sim 1$ in (2.1). We have thus established that the condition of a negligible loss of kinetic energy uniquely determines the exponents p, q .

In Childress (2008) we described the solution to the variational problem for conserved energy and volume. Details, and related work without energy conservation, are given in Childress (2009). The form of the optimizing dipole is shown in figure 1. The extruded ‘tail’ conserves volume while negligibly reducing kinetic energy.

Despite their origin from a problem with axial symmetry, these last estimates provide a crucial piece of information concerning the structure of fast-growing vortical structures of this kind. For an anti-parallel, symmetric pair of adjacent

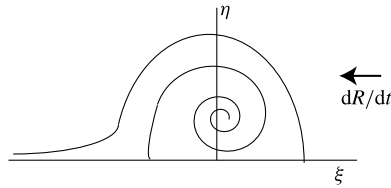


FIGURE 2. The upper half of the snail, showing the instantaneous flow lines relative to a co-moving frame, in local coordinates $(\xi, \eta) = R^{3/4}(x, y)$. The spiral shown is the unique interior flow line terminating at a stagnation point on $\eta = 0$. Note that the streamfunction is multivalued on this spiral, jumping across it by an amount equal to the flux into the tail, as a result of the apparent distribution of sources across the dipole. In the asymptotics to be developed the spiral is much ‘tighter’ than depicted here.

eddies, conservation of energy forces a contraction of eddy cross-section over and above that imposed by conservation of volume. Relative to a co-moving frame with coordinates suitably normalized (here by a factor $R^{3/4}$), the apparent flow now contains a small non-solenoidal component, effectively feeding volume into the tail as the true cross-section contracts. The key point is that this component breaks the constraint of closed streamlines that prevails without it. In effect conservation of energy turns a structurally unstable topology into a structurally stable, spiral topology. We sketch the proposed flow lines of the upper eddy, a structure we shall refer to as the ‘snail’, in figure 2. (Note that the direction of motion of the vortices is opposite to the direction of crawling of the ‘snail’.)

Since we shall not consider here the calculation of the ‘optimizer’ shown in figure 1, it is perhaps useful to explain how it relates to the construction to follow. To obtain the fastest growth of vorticity under the constraint of constant kinetic energy we maximize the velocity at a ‘target’ vortex ring by arranging the available vorticity in an optimal way. Since we are interested only in the maximal final rate of growth at large times, we examine the maximum absolute value k of ω divided by distance R from the axis of symmetry. The $\pm k$ is used in eddies of uniform vorticity, and the shape of the dipole chosen to maximize the velocity of the target ring. Since at large R we know that vorticity on every ring is proportional to R , we may bound the growth of vorticity as $t \rightarrow \infty$. It is shown in Childress (2009) that for R large compared to the diameter of the dipole cross-section, the optimizer satisfies $|\omega| \leq Ck^{5/3}E^{1/3}t^{4/3} + O(t^{1/3})$, where C is a positive number and E is the total kinetic energy of the toroidal dipole structure (with unit density). For the Sadovskii snail R would be chosen so large that the asymptotic Sadovskii state had been reached, and so a similar estimate, applied for the uniform vorticity eddies, would apply. But the number C would presumably be smaller. We have not computed this number since the exponent $4/3$ is the only property of interest here.

2.2. The change of topology in AFWOS

We now summarize results to be derived in the next section. An asymptotic analysis of the maximal growth of vorticity in time results in the leading-order flow and vorticity field ω taking the form of an arbitrary 2-D eddy with closed streamlines. At the next order, however, contour averaging introduces a compatibility constraint on the leading vorticity term, needed to ensure the existence of the solution at second order.

We are dealing, therefore, with a singular perturbation in the topology of the flow. Realizing that the actual structure at second order is that of the snail, we see that

the compatibility constraints on individual closed streamlines are removed, provided that the perturbed velocity field is taken as the ‘leading flow field’, establishing the spiral flow lines. Thus we have an example of bringing forward a second-order effect, in order to completely reorder a calculation, here for the purpose of correctly identifying the spiral topology. Once this is done, contour integration along spiral flow lines terminating in the tail determines the tail, that is, determines the vorticity shed to the wake, irrespective of the particular form of the forcing at second order.

Analysis of the snail configuration will lead naturally to the hypothesis that the preferred ultimate vorticity distribution is one which is piecewise constant, corresponding to zeroth-order eddies which are of the form of the Sadovskii vortex. We use ‘the’ here to refer to the special case where velocity (and the total pressure) is continuous at the dipole boundary. This dipole is one of a family, allowing a discontinuity of velocity at the boundary, considered by Sadovskii (1971). The solution of interest here was independently studied by Saffman & Tanveer (1982); for a review of these problems see Moore, Saffman & Tanveer (1988). The constant vorticity regions emerge from any other dipolar configuration by the stripping away of vorticity into the tail, leaving tubular neighbourhoods of two symmetric anti-parallel vortex lines, thus giving the Sadovskii structure.

3. Analysis of vorticity growth in AFWOS

We now turn to the asymptotic analysis of a dipolar vortex structure under the constraints of AFWOS, for large values of the dipole radius R . Motivated by the preceding estimates and bounds, we shall seek a structure whose cross-sectional area decreases as $R^{-3/2}$ but which maintains self-similarity of shape to leading order. We shall not initially fix p to be $3/4$, since the arguments in this section are kinematic in nature. In contrast to the analysis to be presented in §4, we here consider a slightly more general class where area goes as R^{-2p} , and impose explicitly the scaling associated with $R \sim t^{4/3}$ and a dipole area decreasing as t^{-2} . We then assess the resulting equations for large t .

3.1. Local analysis of eroding dipoles

In cylindrical coordinates (r, z, θ) (this order being chosen as we shall be working largely in the (r, z) -plane), the vorticity equation is

$$\left[\frac{\partial}{\partial t} + u_r \frac{\partial}{\partial r} + u_z \frac{\partial}{\partial z} \right] \frac{\omega}{r} = 0, \quad \omega = \frac{\partial u_z}{\partial r} - \frac{\partial u_r}{\partial z}. \quad (3.1)$$

The conservation of volume is expressed, for an incompressible fluid of unit density, by

$$\frac{1}{r} \frac{\partial r u_r}{\partial r} + \frac{\partial u_z}{\partial z} = 0. \quad (3.2)$$

We now pass to local coordinates (x, y) though the transformation $r = R(t) + x$, $z = y$, $u_r = \dot{R} + u$, $u_z = v$, with $\dot{R} = dR/dt$. We then have

$$\left[\frac{\partial}{\partial t} + u \frac{\partial}{\partial x} + v \frac{\partial}{\partial y} \right] \frac{\omega}{R(t) + x} = 0, \quad \omega = v_x - u_y, \quad (3.3)$$

$$u_x + v_y + \frac{u}{R(t) + x} = - \frac{\dot{R}}{R(t) + x}. \quad (3.4)$$

The time derivative is now understood to be for x fixed.

Writing

$$\omega = \omega_0 \frac{R(t)}{R_0} \Omega(x, y, t), \tag{3.5}$$

where R_0 is a reference length and ω_0 a reference vorticity, for example associated with initial conditions, we have

$$\left[\frac{\partial}{\partial t} + u \frac{\partial}{\partial x} + v \frac{\partial}{\partial y} \right] \frac{\Omega R}{R+x} = 0, \quad u_x + v_y + \frac{u}{R+x} = -\frac{\dot{R}}{R+x}. \tag{3.6a,b}$$

Next, let $a(t) = R_0(R(t)/R_0)^{-p}$ be a lateral scale for the dipole, and set $(\xi, \eta) = a^{-1}(x, y)$. Here p is an exponent we expect to be $3/4$ from the previous section, but we leave it unspecified for the moment as we wish to emphasize the independence of the constraint of conservation of energy from the form of the dipole topology. We may assume for eroding vortices that $p > 1/2$ (so volume a^2R decreases). Lastly, we set

$$(u, v) = \omega_0 R_0 \left(\frac{R}{R_0} \right)^{1-p} (U, V) - p \left(\frac{\dot{R}}{R} \right) (x, y), \tag{3.7}$$

and define a dimensionless time τ by

$$\frac{\partial \tau}{\partial t} = \omega_0 \frac{R}{R_0}. \tag{3.8}$$

In these variables we set $h = 1 + x/R$ and have

$$\left[\frac{\partial}{\partial \tau} + U \frac{\partial}{\partial \xi} + V \frac{\partial}{\partial \eta} \right] \frac{\Omega}{h} = 0, \quad U_\xi + V_\eta + \frac{\varepsilon U}{h} = \frac{(2p-1)\varepsilon}{h}. \tag{3.9a,b}$$

Here we have chosen ω_0, R_0 so that

$$\dot{R} = \omega_0 R_0 (R/R_0)^{1-p}, \quad R = R_0 (1 + p\omega_0 t)^{1/p}, \tag{3.10a,b}$$

and therefore

$$\varepsilon = \left(\frac{R_0}{R} \right)^{1+p} = \frac{\dot{R} R_0}{\omega_0 R^2} = \frac{R_\tau}{R} = \varepsilon(\tau) = \frac{1}{1 + (1+p)\tau} \sim (1+p)^{-1} \tau^{-1}, \quad h = 1 + \varepsilon \xi. \tag{3.11}$$

One useful point to note is that the tail thickness here is $R^{-(1+2p)}$ and so the vorticity in the tail contributes a velocity of order $R \times R^{-(1+2p)} \sim R^{-2p}$, whereas relative to the head the fluid flow exits the tail with velocity of order R^{1-p} so as to match with the free stream velocity, consistent with the above estimates. Thus the vorticity in the tail contributes a velocity which is negligible compared to the free stream.

3.2. Formal expansion

We now return to (3.9) and carry out a formal expansion in ε (or τ^{-1}) for the flow in the upper half of the dipole. We introduce the new time variable τ^* by

$$\frac{\partial \Omega_0}{\partial \tau} = \varepsilon \frac{\partial \Omega_0}{\partial \tau^*}. \tag{3.12}$$

Thus $\tau^* = (1+p)^{-1} \log[1 + (1+p)\tau] \sim (1+p)^{-1} \log \tau$. Then (3.9) becomes

$$\left[\varepsilon \frac{\partial}{\partial \tau^*} + U \frac{\partial}{\partial \xi} + V \frac{\partial}{\partial \eta} \right] \frac{\Omega}{h} = 0, \quad U_\xi + V_\eta + \frac{\varepsilon U}{h} = \frac{(2p-1)\varepsilon}{h}. \quad (3.13a,b)$$

The reasoning here is that the scaling of the coordinates and velocity components has already absorbed the dominant time dependence, and we are left with the slower dependence on τ^* . Of course now ε may be regarded as a function of τ^* , with

$$\frac{\partial \varepsilon}{\partial \tau^*} = -(p+1)\varepsilon. \quad (3.14)$$

We now let $\mathbf{Q} \equiv (U, V) = \mathbf{Q}_0(\xi, \eta, \tau^*) + \varepsilon \mathbf{Q}_1(\xi, \eta, \tau^*) + \dots$ and $\Omega = \Omega_0(\xi, \eta, \tau^*) + \varepsilon \Omega_1(\xi, \eta, \tau^*) + \dots$. We then obtain the equations

$$\mathbf{Q}_0 \cdot \nabla \Omega_0 = 0, \quad \nabla \cdot \mathbf{Q}_0 = 0, \quad (3.15)$$

$$\frac{\partial \Omega_0}{\partial \tau^*} + \mathbf{Q}_0 \cdot \nabla \Omega_1 + \mathbf{Q}_1 \cdot \nabla \Omega_0 - U_0 \Omega_o = 0, \quad \nabla \cdot \mathbf{Q}_1 = 2p - 1 - U_0. \quad (3.16)$$

Let us first solve (3.15) simply by setting $\Omega_0 = \text{constant}$ in a lobe of the vortex. This is a special case which, however, will be shown in the following subsection to be the only allowed solution for the eroding dipole. At next order a particular solution is seen to satisfy

$$\Omega_1 = \frac{\partial V_1}{\partial \xi} - \frac{\partial U_1}{\partial \eta} = \Omega_0 \xi, \quad \frac{\partial U_1}{\partial \xi} + \frac{\partial V_1}{\partial \eta} = 2p - 1 - U_0. \quad (3.17a,b)$$

For example, we can take

$$U_1 = \frac{1}{2} \eta V_0 + (p - \frac{1}{2}) \xi, \quad V_1 = (p - \frac{1}{2}) \eta + \frac{1}{2} \Omega_0 \xi^2 + \frac{1}{2} (\Psi_0 - \eta U_0) - \frac{1}{4} \eta^2 \Omega_0, \quad (3.18a,b)$$

where the streamfunction Ψ_0 is specified by

$$(U_0, V_0) = \left(-\frac{\partial \Psi_0}{\partial \eta}, \frac{\partial \Psi_0}{\partial \xi} \right). \quad (3.19)$$

Any potential flow can be added to this solution and the result matched with an exterior potential flow to make the velocity continuous on the bounding streamline of the vortex.

The point is then that we have a way of extending the zeroth-order solution. In fact we know that there exists a \mathbf{Q}_0 of the desired form, namely the Sadovskii vortex with continuous total pressure.

3.3. The general case

We shall say that the dipole vortex is compatible if an asymptotic solution exists inclusive of the terms of order ε . We now establish a simple but somewhat surprising result:

LEMMA 1. *The class of eroding dipoles just studied, where vorticity is constant in each eddy, is the only compatible, zeroth-order flow field independent of τ^* .*

To prove this we note from (3.15) that the general solution has the form $\Omega_0 = F(\Psi_0)$ where F is an arbitrary function. Then, from (3.16) we see that

$$\mathbf{Q}_0 \cdot \nabla \Omega_1 + \mathbf{Q}_1 \cdot \nabla \Psi_0 F'(\Psi_0) = U_0 F(\Psi_0). \tag{3.20}$$

We now restrict attention to one of the two regions of closed streamlines of the zeroth-order dipole, and introduce the contour average

$$\langle \cdot \rangle = \oint \frac{\cdot}{|\mathbf{Q}_0|} ds, \tag{3.21}$$

taken along the direction of flow around a streamline of the flow \mathbf{Q}_0 in the upper eddy. Then it is easy to see from (3.20) that

$$\langle \mathbf{Q}_0 \cdot \nabla \Omega_1 \rangle = 0 = F'(\Psi_0) \oint \mathbf{Q}_1 \cdot \mathbf{n} ds + F(\Psi_0) \langle U_0 \rangle. \tag{3.22}$$

However, from the divergence theorem and (3.16),

$$\oint \mathbf{Q}_1 \cdot \mathbf{n} ds = \int \int (2p - 1 - U_0) d\xi d\eta = (2p - 1) A(\Psi_0) \tag{3.23}$$

(U_0 makes no contribution), where A is area within a contour of constant Ψ_0 in the (ξ, η) plane. Also

$$\langle U_0 \rangle = \oint \frac{U_0}{|\mathbf{Q}_0|} ds = \oint \frac{dx}{ds} ds = 0. \tag{3.24}$$

It then follows from (3.22) and $p > 1/2$ that $F'(\Psi_0) = 0$ and the lemma is proved.

This lemma brings to mind the classical Prandtl–Batchelor result concerning the constancy of vorticity in steady flow in a region of closed streamlines at large Reynolds number; see Batchelor (1956). Indeed, that work inspired investigation of the associated Euler flows (Childress 1966). However, the proof of the Prandtl–Batchelor theorem uses the Navier–Stokes equation, since the result depends upon the small but persistent diffusion of vorticity, whereas here the drift to the constant state follows from erosion of vorticity in an inviscid flow and thus convergence to the vorticity value at the vortex centre.

Now we allow Ω_0 to depend upon τ^* . The contour average then gives

$$\left\langle \frac{\partial \Omega_0}{\partial \tau^*} \right\rangle + \frac{\partial \Omega_0}{\partial \Psi_0} \oint \mathbf{Q}_1 \cdot \mathbf{n} ds + F(\Psi_0) \langle U_0 \rangle = \left\langle \frac{\partial \Omega_0}{\partial \tau^*} \right\rangle + \frac{\partial \Omega_0}{\partial \Psi_0} (2p - 1) A(\Psi_0). \tag{3.25}$$

But

$$\left\langle \frac{\partial \Omega_0}{\partial \tau^*} \right\rangle = \frac{\partial \Omega_0}{\partial \tau^*} \Big|_{\Psi_0} \langle 1 \rangle + \frac{\partial \Omega_0}{\partial \Psi_0} \left\langle \frac{\partial \Psi_0}{\partial \tau^*} \right\rangle. \tag{3.26}$$

Since (see e.g. Childress (1987))

$$\langle 1 \rangle = \frac{\partial A}{\partial \Psi_0}, \quad - \left\langle \frac{\partial \Psi_0}{\partial \tau^*} \right\rangle = \frac{\partial A}{\partial \tau^*}, \tag{3.27a,b}$$

we have

$$\frac{\partial \Omega_0}{\partial \tau^*} \frac{\partial A}{\partial \Psi_0} - \frac{\partial \Omega_0}{\partial \Psi_0} \left[\frac{\partial A}{\partial \tau^*} - (2p - 1) A \right] = 0. \tag{3.28}$$

Consequently

$$\Omega_0 = G(e^{-(2p-1)\tau^*} A(\Psi_0, \tau^*)), \tag{3.29}$$

for some function G , But $e^{-(2p-1)\tau^*} \sim \tau^{-(2p-1)/(1+p)} \sim R^{1-2p}$ and $R^{1-2p}A$ is equal to R times the dimensional area. Thus we obtain a steady flow-preserving constant total volume, with area shrinking as R^{-1} . This is a compatible dipole for arbitrary $F(\Psi_0)$ since it corresponds to (3.22), (3.23) with $p = 1/2$. The dependence on τ^* when $p > 1/2$ results from observing a steady volume-preserving structure within coordinates shrinking faster than is required by conservation of volume.

Using the term ‘steady’ in the above sense, meaning that Ω_0 is independent of τ^* , we thus have the following result:

THEOREM 1. *The compatible dipole vortex structures consist of the steady eroding vortices ($p > 1/2$) with piecewise constant vorticity and the steady volume-preserving vortices ($p = 1/2$) with arbitrary $F(\Psi_0)$.*

We emphasize that compatibility is a fairly weak measure of dynamic consistency, leaving the requirement of constant kinetic energy as an added and independent constraint. The exponent p needs to be fixed by a full asymptotic solution for large R involving matching an eroding vortex to an external potential flow, as well as proper treatment of the vorticity ‘tail’, and this requires a numerical solution for the perturbed Sadovskii vortex, a problem we take up in the next section. We know of course that the unique compatible structure preserving total kinetic energy is the steady eroding vortex with $p = 3/4$.

In spite of the limited implications of compatibility, we do gain a basic constraint of the zeroth-order structure. We know that the vorticity squared of the dipole, times the area of one vortex, divided by the speed of propagation squared, must equal 37.11 (Saffman & Tanveer 1982). Let the dipole be at position $R \gg R_0$, moving with speed $\dot{R} = \omega_0 R_0 (R/R_0)^{1/4}$, and having vorticity $\omega_0 R/R_0$ and area $2A(R_0/R)^{3/2}$. It then follows that

$$R_0 = \sqrt{A/37.11} \tag{3.30}$$

is our reference length.

We remark that the structure of our preferred dipole with $p = 3/4$ can be studied directly in the stable topology. The idea is simply to take the ‘zeroth-order’ term of the snail velocity field, (U_s, V_s) say, to include the apparent fluid source to order ε :

$$(U_s, V_s) = (U_0, V_0) + \frac{1}{4}\varepsilon(\xi, \eta), \quad \frac{\partial U_0}{\partial \xi} + \frac{\partial V_0}{\partial \eta} = 0, \tag{3.31a,b}$$

where again (U_0, V_0) is the unperturbed Sadovskii velocity field. Our ‘zeroth-order’ problem then becomes;

$$U_s \frac{\partial \Omega_s}{\partial \xi} + V_s \frac{\partial \Omega_s}{\partial \eta} = 0, \quad \Omega_s = \frac{\partial V_s}{\partial \xi} - \frac{\partial U_s}{\partial \eta}. \tag{3.32a,b}$$

Our result is now immediate. All flow lines of each vortex are spirals out of a common centre. Since Ω_s is constant on these flow lines, Ω_s must be equal everywhere to the value at this centre. We thus may understand the perturbed Sadovskii structure as a result of eroding away the outer layers of the initial structure.

3.4. Summary of the composite solution at leading order

We have seen that the snail emerges as the only compatible vortex structure conserving kinetic energy. We now shall describe the ‘leading-order’ structure in its entirety, including the external potential flow. By ‘leading order’ we here mean that the nominally higher-order effect needed to capture the vortex shrinkage is to be included as a leading-order effect. We thus will describe a perturbed Sadovskii vortex. We are here neglecting entirely the dynamics of erosion. We assume a contracting Sadovskii dipole, at a rate determined by the assumption of energy conservation, and match this with an external flow. To justify this leading-order solution one must derive the erosion from the equations of motion, and this problem we will take up in the following section.

3.4.1. The exterior flow

We begin with calculation of a uniform approximation to the external potential flow. This flow exists outside the structure consisting of the Sadovskii vortex plus tail. In fact the tail will not be considered in detail as it will have no active role in the leading-order solution.

It is helpful to first consider a simpler potential flow problem, that of an expanding, volume-preserving torus centred at $r = R(t)$ with local radius $a(t)$. We present this calculation in appendix A. It will suffice here to give the result obtained for the velocity potential ϕ in the immediate neighbourhood of the torus:

$$\phi_{\text{torus}} = a^2 \dot{R} \left[-\frac{x}{\rho^2} + \frac{1}{2R} \log \frac{8R}{\rho} + \frac{1}{2R} \frac{x^2}{\rho^2} \right] + O(a^2 \dot{R}/R^2). \tag{3.33}$$

Here the notation is essentially that used earlier in § 2.1 with $\rho^2 = x^2 + y^2$. Note that, relative to an observer moving with the torus, the normal velocity on $\rho = a$ is \dot{a} , as required. Also

$$\int_0^{2\pi} 2\pi a(R + a \cos \theta) \left. \frac{\partial \phi}{\partial \rho} \right|_{\rho=a} d\theta = 0, \tag{3.34}$$

consistent with volume conservation. Moreover, if we wish to create a potential flow which conserves kinetic energy, with the cross-sectional area decreasing as $R^{-3/4}$, we need only change the middle term of (3.33) to $(3/4R) \log(8R/\rho)$, with a corresponding addition of a multiple of (A 1) to the potential function.

Now we can obtain (3.33) directly by observing that $\phi_0 \equiv -a^2 \dot{R} x/\rho^2$ is the perturbed potential for 2-D flow past a cylinder. In 3-D we need to solve

$$\phi_{xx} + \phi_{yy} = \nabla^2 \phi = -\frac{1}{R+x} \phi_x, \tag{3.35}$$

and with $\phi = \phi_0 + \phi_1 + \dots$ we would have $\nabla^2 \phi_1 = -R^{-1} \partial \phi_0 / \partial x$. Thus

$$\phi_1 = -\frac{1}{2R} x \phi_0 \tag{3.36}$$

plus a harmonic function. The latter must be proportional to $\log \rho$ in order to satisfy (up to a function of time) (3.34), yielding (3.33). The terms involving R^{-1} come from the shrinking of the cross-section as the torus expands, and the effect of curvature of

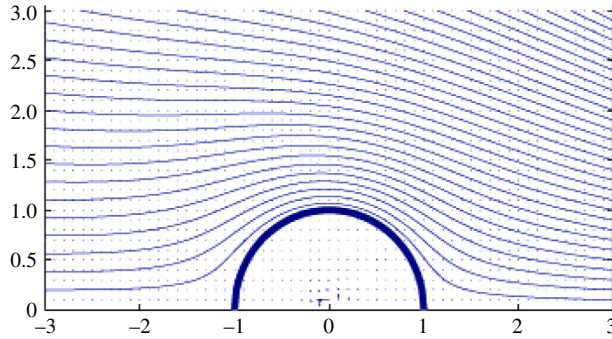


FIGURE 3. (Colour online) Flow lines for the velocity field with potential given by (3.37) with $1/R = 0.2$.

the axis of the torus. We show in figure 3 a plot of the flow lines corresponding to the potential

$$\phi_{\text{torus}} = -x - \frac{x}{\rho^2} - \frac{1}{2R} \log \rho + \frac{1}{2R} \frac{x^2}{\rho^2} + \frac{1}{4R} (x^2 + y^2), \quad (3.37)$$

where we have added a dilation to make the normal velocity vanish on $\rho = 1$.

For the Sadovskii dipole (not the snail, so dipole volume is preserved) we can proceed similarly. Let ϕ_0 be the dipole's 2-D exterior flow. Then

$$\phi = \phi_0 - \frac{1}{2R} x\phi_0 + \phi_1 + \dots, \quad (3.38)$$

where ϕ_1 is a harmonic function, which sets the appropriate normal velocity at the boundary of the dipole; see below. In the following subsection we match (3.38), modified to produce the snail, to a potential flow defined on the scale of the toroidal dipole, using the functions developed in appendix A.

3.4.2. A uniformly valid solution at leading order

Our aim now is to exhibit a uniformly valid and compatible dipole to leading order in the sense that the first-order terms necessary to describe the topology of the flow lines are included. We know that the zeroth-order dipole is the Sadovskii vortex, and what follows is an approximate treatment of the modifications which produce the snail.

Let \tilde{S} denote the cross-section of the dipole in (ξ, η) coordinates. Referring back to the coordinates in §3.1, we define S by

$$(x, y) \in S \Leftrightarrow (\xi, \eta) \in \tilde{S}. \quad (3.39)$$

Then in the vicinity of the dipole, including both interior and exterior, the solution we seek has the form

$$\mathbf{u} = -\dot{R}\mathbf{i} + \mathbf{u}_{\text{dipole}} + \mathbf{u}_{\text{shrink}}. \quad (3.40)$$

The first two terms on the right-hand side of (3.40) describe the instantaneous flow for the Sadovskii dipole relative to the co-moving coordinates, and in the exterior we include the correction in (3.38) associated with the 'squeeze' flow. The area of the

dipole cross-section is $\mathcal{A} \equiv A(R_0/R)^{3/2}$. In the intermediate region $\mathcal{A} \ll x^2 + y^2 \ll R^2$ we will have

$$\mathbf{u}_{dipole} \sim -\frac{k_1 \dot{R}}{\pi} \mathcal{A} \nabla \left[\frac{x}{\rho^2} - \frac{1}{2R} \frac{x^2}{\rho^2} \right], \quad \rho^2 = x^2 + y^2, \quad (3.41)$$

for some positive constant k_1 .

The term \mathbf{u}_{shrink} represents the potential flow due to dipole shrinkage. Thus $\mathbf{u}_{shrink} = \nabla \phi_{shrink}$ outside the dipole, where in the intermediate region we may write

$$\phi_{shrink} = -\frac{k_2 \dot{R}}{\pi} \frac{\mathcal{A}}{R} \log(8R/\rho) + \phi_{add}, \quad (3.42)$$

where k_2 is a constant to be determined and ϕ_{add} is harmonic and $O(\rho^{-2})$ for large ρ . Note that within the dipole the shrinkage of the snail in the (x, y, z) frame is a uniform contraction equivalent to that of a volume-preserving torus (vorticity grows as if there were no erosion), but the boundary of the dipole contracts faster owing to the stripping away of vorticity.

Other first-order terms are ignored. For example the sink distribution contained in \mathbf{u}_{shrink} is over S , instead of the perturbed vortex which includes the tail. There is thus a ‘boundary layer’ of vorticity missing here, associated with a tangential jump in velocity within first-order terms. Also the tangential component of the perturbation flow in the exterior has not been matched to an interior flow perturbation. To put this another way, (3.40) captures the shrinking snail, but makes small errors in its shape.

Let us now consider the exterior potential flow relative to the fluid at infinity. Given the instantaneous centre curve of the Sadovskii vortex, we surround the structure by a concentric torus of cross-sectional area large compared to the Sadovskii vortex area but small compared to R^2 . Call the surface of this torus ∂T . In the region outside of ∂T we shall represent the potential of the flow relative to the fluid at infinity in the form

$$\phi_{ext} = -2\dot{R}\mathcal{A}(k_1\Phi + k_3\phi/R), \quad (3.43)$$

where Φ, ϕ are as given in appendix A, and k_3 is another constant to be determined. On ∂T we have, to leading order

$$\phi_{ext} \sim \frac{k_1}{\pi} \dot{R}\mathcal{A} \left[-\frac{x}{\rho^2} - \frac{1}{2R} \log \frac{8R}{\rho} + \frac{1}{2R} \frac{x^2}{\rho^2} \right] + \frac{k_3 \dot{R}\mathcal{A}}{\pi R} \log \frac{8R}{\rho}. \quad (3.44)$$

Using (3.44) we may compute the net flux of fluid into T , which must equal the rate of change of volume of the toroidal dipole (and the flux into the tail). We obtain

$$-4\pi \dot{R}\mathcal{A}(-k_1 + k_3) = \frac{d}{dt} 2\pi R\mathcal{A} = \frac{d}{dt} 2\pi R A \left(\frac{R_0}{R} \right)^{3/2} = -\pi \dot{R}\mathcal{A}. \quad (3.45)$$

Thus

$$k_3 = \frac{1}{4} + k_1. \quad (3.46)$$

On the other hand, approaching ∂T from within we may write,

$$\mathbf{u} \sim \nabla \phi_{in}, \quad (3.47)$$

where

$$\phi_{in} \sim \frac{k_1}{\pi} \dot{R} \mathcal{A} \left[-\frac{x}{\rho^2} + \frac{1}{2R} \frac{x^2}{\rho^2} \right] - \frac{\dot{R} \mathcal{A}}{\pi R} k_2 \log(8R/\rho). \quad (3.48)$$

Comparing (3.44) and (3.48) we see that

$$k_2 = \frac{1}{2}k_1 - k_3, \quad k_2 = -\frac{1}{2}k_1 - \frac{1}{4}. \quad (3.49a,b)$$

Here the term $-(1/2)k_1$ contributes a correction to \mathbf{u}_{dipole} to yield zero flux into the dipole, while the term $-1/4$ gives the flux into the dipole, flux which is then expelled into the tail.

Since we have correctly established the flow through ∂T we are assured that the boundary condition in ∂S can be met for a suitable ϕ_{add} in (3.42), so as to match the normal velocity associated with shrinkage due to geometry and erosion.

3.4.3. Summary

We recapitulate the results of this section in anticipation of the redevelopment of the problem in the next section. We have established that the imposition of local conservation of energy leads to an eroding Sadovskii dipole. From this we deduce the existence of a tail to which the eroded vorticity is extruded. However, if correct this model should evolve naturally from the dynamics. In particular the scaling following from $p = 3/4$ should evolve as an eroding structure with locally steady structure in the shrinking coordinates, and the flow of vorticity into the tail should be a derivable perturbation of the underlying Sadovskii eddy. It is just such a dynamical calculation that we now want to pursue.

This will entail a somewhat different approach in the coordinates used and the formulation of the underlying scaling of the dipole as an unknown. While this will involve a more systematic asymptotic theory, we will again encounter elements of the solution already exhibited. For example the terms $\xi \Omega_0$ and $-U_0$ on the right-hand side in (3.17) embody the curvature of the cylindrical geometry, the former closely related to terms in § 4 indicated by the superscript ‘sq’, short for ‘squeeze’. These terms arise from curvature of the vortex lines, which induces a flow along the binormal, causing curved anti-parallel vortex filaments to be squeezed together. This is a main cause of shedding of vorticity into the tail.

4. Full analysis of the perturbed Sadovskii dipole

We now determine dynamically how the perturbation of the Sadovskii dipole shape leads to erosion of vorticity and therefore determines the rate of shrinkage and speed of the dipole. The parameter p in § 2 becomes an unknown to be determined from an asymptotic solution of Euler’s equations valid for large a/R . We will find that p may be computed numerically from the condition that a scaling actually exists, i.e. that in suitable coordinates the structure appears steady, just as to leading order the snail is steady in local (ξ, η) coordinates. We shall maintain a certain part of the notation of the previous sections; however, there will be departures and so the reader should regard this section as largely self-contained in notation.

4.1. Inner expansion about a steady Sadovskii vortex

We seek a solution for the vortex pair evolution at large radii $R(t)$ for which it is helpful to set $r\varpi = \omega = \partial u_z / \partial r - \partial u_r / \partial z$ and to solve (3.2) using a Stokes

streamfunction ψ , $u_r = -r^{-1}\partial\psi/\partial z$, $u_z = r^{-1}\partial\psi/\partial r$. Introducing these into (3.2), we seek to solve the vorticity equation

$$\frac{\partial\varpi}{\partial t} + \frac{1}{r} \frac{\partial(\psi, \varpi)}{\partial(r, z)} = 0, \quad \varpi = \frac{1}{r} \frac{\partial}{\partial r} \left(\frac{1}{r} \frac{\partial\psi}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2\psi}{\partial z^2}. \tag{4.1a,b}$$

This simplifies a little if we replace the radial coordinate r by $(1/2)r^2$ and this motivates the change of variables from (r, z, t) to (ξ, η, τ) given by

$$r^2 = R^2 + 2aR\xi = R_0^2(g^2 + 2fg\xi), \quad z = a\eta = R_0f\eta, \quad d\tau/dt = \omega_0R/R_0 = \omega_0g, \tag{4.2a-c}$$

with R_0 and ω_0 dimensional reference quantities as before. We have introduced dimensionless radii given by $a(t) = R_0f(\tau)$ and $R(t) = R_0g(\tau)$. The transformation differs only in minor ways from that introduced earlier in §3.1. For the fields we set

$$\varpi(r, z, t) = \frac{\omega_0}{R_0} \tilde{\varpi}(\xi, \eta, \tau), \quad \psi(r, z, t) = \omega_0R_0^3f^2g^2\tilde{\psi}(\xi, \eta, \tau). \tag{4.3a,b}$$

Dropping any tildes leaves the vorticity equation and vorticity–streamfunction link as

$$\frac{\partial\varpi}{\partial\tau} - \frac{\dot{g}}{f} \frac{\partial\varpi}{\partial\xi} - \frac{\dot{g}}{g} \xi \frac{\partial\varpi}{\partial\xi} - \frac{\dot{f}}{f} \left(\xi \frac{\partial\varpi}{\partial\xi} + \eta \frac{\partial\varpi}{\partial\eta} \right) + \mathcal{J}(\psi, \varpi) = 0, \tag{4.4}$$

$$\varpi = \frac{\partial^2\psi}{\partial\xi^2} + \frac{1}{1+2fg^{-1}\xi} \frac{\partial^2\psi}{\partial\eta^2}, \tag{4.5}$$

where we use \mathcal{J} for a Jacobian with respect to the (ξ, η) coordinates and a dot (in this section only) for a τ -derivative of f or g . This formulation is exact but we have in mind $g = R/R_0 \gg 1$ and $f = a/R_0 \ll 1$ for large times and that these are slowly varying, that is,

$$f \ll 1, \quad g \gg 1, \quad \dot{f}/f \ll 1, \quad \dot{g}/g \ll 1; \tag{4.6a-d}$$

these may be verified *a posteriori*. We remark that the assumption $p = 3/4$ in §2 leads to $f \sim t^{-1}$, $g \sim t^{4/3}$, so that, as before, an expansion for large t is implied. Our aim now is to obtain a value for p asymptotically by analysis of the shedding of vorticity into the tail.

Thus for an inner solution, that is valid for $(\xi, \eta) = O(1)$, we drop the $2fg^{-1}\xi$ term in (4.5) and the frame contraction terms involving \dot{f}/f and \dot{g}/g in (4.4) at leading order. We use an inner expansion

$$\varpi = \varpi_0 + \frac{f_0}{g_0} \varpi_1 + \dots, \quad \psi = \psi_0 + \frac{f_0}{g_0} \psi_1 + \dots, \quad g = g_0 + g_1 + \dots, \quad f = f_0 + f_1 + \dots, \tag{4.7a-d}$$

in which we will find that the ϖ_1 and ψ_1 are of order unity. This gives, at leading order, equations for purely 2-D Euler flow,

$$\frac{\partial\varpi_0}{\partial\tau} - c_0 \frac{\partial\varpi_0}{\partial\xi} + \mathcal{J}(\psi_0, \varpi_0) = 0, \quad \varpi_0 = \frac{\partial^2\psi_0}{\partial\xi^2} + \frac{\partial^2\psi_0}{\partial\eta^2}, \tag{4.8a,b}$$

with c_0 defined by

$$c_0 = \dot{g}_0/f_0. \tag{4.9}$$

Now all we have done so far is valid for any $f_0(\tau)$ and $g_0(\tau)$ and so without further information c_0 could also depend on τ . However we are seeking a leading-order approximation as the steady Sadvskii vortex with continuous velocity. We thus set the vorticity, streamfunction and speed, that is (ϖ_0, ψ_0, c_0) , to be one of the family of such vortices, with c_0 taken as constant. We will later choose one with $\varpi_0 = \pm 1$ in the two lobes, and $c_0 = 1$, but for the moment the choice is arbitrary. Thus, (4.9) provides a single ordinary differential equation (ODE) linking f_0 and g_0 ; we need a further ODE to close the system, and this will emerge at the next order.

Although the choice of our leading-order steady solution is arbitrary, the fact that it is one of a family has important implications. It means that an infinitesimal translation,

$$\varpi_0^{trans} = \frac{\partial \varpi_0}{\partial \xi}, \quad \psi_0^{trans} = \frac{\partial \psi_0}{\partial \xi}, \tag{4.10a,b}$$

satisfies the linear equations

$$-c_0 \frac{\partial \varpi_0^{trans}}{\partial \xi} + \mathcal{J}(\psi_0^{trans}, \varpi_0) + \mathcal{J}(\psi_0, \varpi_0^{trans}) = 0, \quad \psi_0^{trans} = \mathcal{G} \varpi_0^{trans}. \tag{4.11}$$

For a solution (ϖ_0, ψ_0, c_0) , a rescaled solution is $(\varpi_0(\lambda\xi, \lambda\eta), \lambda^{-2}\psi_0(\lambda\xi, \lambda\eta), \lambda^{-1}c_0)$ for any λ . Thus, taking the derivative with respect to λ at $\lambda = 1$, we obtain a solution giving an infinitesimal change of scale

$$\varpi_0^{scale} = \xi \frac{\partial \varpi_0}{\partial \xi} + \eta \frac{\partial \varpi_0}{\partial \eta}, \quad \psi_0^{scale} = \xi \frac{\partial \psi_0}{\partial \xi} + \eta \frac{\partial \psi_0}{\partial \eta} - 2\psi_0, \tag{4.12a,b}$$

which obeys

$$-c_0 \frac{\partial \varpi_0^{scale}}{\partial \xi} + c_0 \varpi_0^{trans} + \mathcal{J}(\psi_0^{scale}, \varpi_0) + \mathcal{J}(\psi_0, \varpi_0^{scale}) = 0, \quad \psi_0^{scale} = \mathcal{G} \varpi_0^{scale}. \tag{4.13}$$

Here we have introduced \mathcal{G} in (4.11), (4.13) as the operator inverting the Laplacian in (4.8), that is integration against the kernel

$$G(\xi, \eta) = (4\pi)^{-1} \log(\xi^2 + \eta^2) \tag{4.14}$$

in infinite (ξ, η) space.

Having dealt with the leading-order problem we now write down the first-order equation, in which the neglected terms involving $\dot{f}/f, \dot{g}/g$ in (4.4) and $2fg^{-2}\xi$ in (4.5) are reintroduced to drive corrections (ϖ_1, ψ_1) to the fields:

$$\begin{aligned} \frac{\partial \varpi_1}{\partial \tau} - c_0 \frac{\partial \varpi_1}{\partial \xi} - c_1 \frac{\partial \varpi_0}{\partial \xi} - c_0 \xi \frac{\partial \varpi_0}{\partial \xi} + c_0 p_1 \left(\xi \frac{\partial \varpi_0}{\partial \xi} + \eta \frac{\partial \varpi_0}{\partial \eta} \right) \\ + \mathcal{J}(\psi_0, \varpi_1) + \mathcal{J}(\psi_1, \varpi_0) = 0, \end{aligned} \tag{4.15}$$

$$\varpi_1 = \frac{\partial^2 \psi_1}{\partial \xi^2} + \frac{\partial^2 \psi_1}{\partial \eta^2} - 2\xi \frac{\partial^2 \psi_0}{\partial \eta^2}. \tag{4.16}$$

Here we have made use of (4.6) and (4.9), and defined

$$c_1 = \frac{g_0}{f_0^2} (\dot{g}_1 - c_0 f_1), \quad p_1 = -\frac{\dot{f}_0 g_0}{c_0 f_0^2}. \tag{4.17a,b}$$

To deal with this first-order problem, first invert (4.16) as

$$\psi_1 = \mathcal{G}\varpi_1 + \psi_1^{sq}, \quad \psi_1^{sq} \equiv 2\mathcal{G} \left(\xi \frac{\partial^2 \psi_0}{\partial \eta^2} \right) \tag{4.18a,b}$$

and then use (4.10), (4.12) to rearrange (4.15) as

$$\begin{aligned} \frac{\partial \varpi_1}{\partial \tau} - c_0 \frac{\partial \varpi_1}{\partial \xi} + \mathcal{J}(\psi_0, \varpi_1) &= c_0 \xi \frac{\partial \varpi_0}{\partial \xi} - \mathcal{J}(\mathcal{G}\varpi_1, \varpi_0) - \mathcal{J}(\psi_1^{sq}, \varpi_0) \\ &+ c_1 \varpi_0^{trans} - c_0 p_1 \varpi_0^{scale} \end{aligned} \tag{4.19}$$

On the left-hand side we have advection of vorticity ϖ_1 in the basic flow field of the Sadovskii vortex; on the right-hand side are the remaining terms. This equation is ‘driven’ by the terms $c_0 \xi \partial \varpi_0 / \partial \xi$ and $\mathcal{J}(\psi_1^{sq}, \varpi_0)$, in that if these terms were absent a solution would be $\varpi_1 = 0$, $c_1 = p_1 = 0$. Although the driving terms are constant (independent of τ), the solution $\varpi_1(\xi, \eta, \tau)$ will generally not be steady, as it will acquire pieces of ϖ_0^{trans} corresponding to drift in the ξ -direction and ϖ_0^{scale} corresponding to a change in scale; see (4.10)–(4.13).

However we can eliminate these terms by suitable choice of c_1 and p_1 – we will check this numerically in due course – and with this choice we expect to be able to obtain a solution ϖ_1 independent of τ . Note that this choice is available to us as the functions f and g are arbitrary rescalings and so we can choose to fix them order by order. This imposition of a solvability condition gives a solution representing the modified Sadovskii vortex, travelling outwards according to $g_0(\tau)$ and shrinking through shedding vorticity according to $f_0(\tau)$.

So, we suppose we have converged to a steady solution $\varpi_1(\xi, \eta)$ with constants c_1 (which is not of use to us as it involves f_1 as a new unknown) and p_1 which gives a second ODE linking f_0 and g_0 in (4.17). Together with (4.2), (4.9) we obtain

$$f_0 \propto \tau^{p_1/(1+p_1)} \propto (\omega_0 t)^{-1}, \quad g_0 \propto \tau^{1/(1+p_1)} \propto (\omega_0 t)^{1/p_1}, \quad \tau \propto (\omega_0 t)^{1+1/p_1}, \tag{4.20a–c}$$

and we anticipate $p_1 > 0$ so that g_0 increases with t and the approximations are all self-consistent. Here we may identify $p_1 = p$, the exponent introduced in §2.1.

We comment that the term in ψ^{sq} corresponds to the leading-order effect of curved vortex lines creating a flow that drives the two lobes of the Sadovskii vortex together, a weak but controlling effect in our expansion. We should also note that when we invert minus the Laplacian and write down $\psi_0 = \mathcal{G}\varpi_0$ and $\mathcal{G}\varpi_1$, for example in (4.18) we could add on a component which is harmonic in the (ξ, η) plane. In fact fundamentally this is how the distant structure of the vortex would feed into the inner solution, modifying vortex shape and motion. It is clear that the terms that would be incorporated would take the form of a multipole expansion: the first would appear at the level of $\mathcal{G}\varpi_1$ and would correspond to uniform flow. This could be absorbed into c_1 (a Galilean transformation) but would not affect the vortex structure or p_1 .

4.2. Formulation in terms of contours and numerical solution

Now the above is written as if the vorticity fields are smooth, but in fact we are working about the Sadovskii vortex in which the vorticity field is piecewise constant, and so to actually solve the above problem we need to work, not with the fields ϖ_0, ϖ_1 , but instead using contour dynamics. We have in mind here the asymptotic state of the dipole pair for large radius R , where erosion has led to the vorticity being

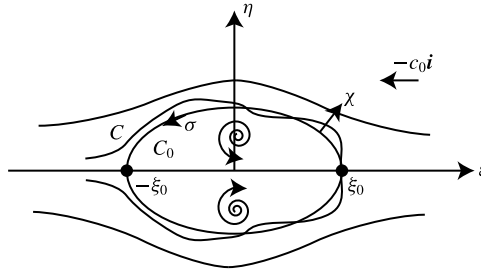


FIGURE 4. Schematic of the perturbed Sadovskii vortex indicating the local Cartesian coordinates (ξ, η) and the curvilinear coordinates (χ, σ) (4.25) adapted to follow the bounding contour C_0 of the unperturbed vortex. The oncoming velocity at infinity is $-c_0 i$.

effectively constant in each lobe and the eroded edge is taken as a discontinuity. We thus need to further manipulate the equations, working in the (ξ, η) plane with the use of polar coordinates (ρ, θ) in this plane when needed. Note that (4.19) takes the form

$$\frac{\partial \varpi_1}{\partial \tau} + \mathbf{U}_0 \cdot \nabla \varpi_1 + \mathbf{U}_1 \cdot \nabla \varpi_0 = 0 \tag{4.21}$$

(we will express \mathbf{U}_0 and \mathbf{U}_1 explicitly below), which is the linear piece of the full equation

$$\frac{D\varpi}{D\tau} \equiv \frac{\partial \varpi}{\partial \tau} + \mathbf{U} \cdot \nabla \varpi = 0, \quad \varpi = \varpi_0 + \varpi_1 + \dots, \quad \mathbf{U} = \mathbf{U}_0 + \mathbf{U}_1 + \dots \tag{4.22}$$

The leading piece of this is $\mathbf{U}_0 \cdot \nabla \varpi_0 = 0$ and gives the steady Sadovskii vortex with the velocity $\mathbf{U}_0 = (U_0, V_0)$ and vorticity linked to the total streamfunction $\Psi_0 = \psi_0 + c_0 \eta$ (including the flow past the vortex) via

$$U_0 = -\frac{\partial \Psi_0}{\partial \eta}, \quad V_0 = \frac{\partial \Psi_0}{\partial \xi}, \quad \varpi_0 = \frac{\partial V_0}{\partial \xi} - \frac{\partial U_0}{\partial \eta}. \tag{4.23a-c}$$

The vortex has vorticity $\varpi_0 = 1$ in a region bounded by the ξ -axis and a contour C_0 in the half-plane $\eta > 0$ and $\varpi_0 = -1$ in the mirror image region; see figure 4. Using the divergence theorem the corresponding streamfunction $\psi_0 = \mathcal{G} \varpi_0$ can be obtained by integrating over the boundaries and gives

$$\begin{aligned} \psi_0(\xi, \eta) = & \frac{1}{4\pi} \int_{C_0} \left\{ \log |(\xi', \eta') - (\xi, \eta)| [(\xi', \eta') - (\xi, \eta)] \cdot (d\eta', -d\xi') \right. \\ & + \log |(\xi', -\eta') - (\xi, \eta)| [(\xi', -\eta') - (\xi, \eta)] \cdot (-d\eta', -d\xi') \left. \right\} \\ & + \frac{\eta}{2\pi} \left[\xi' \log \sqrt{\xi'^2 + \eta^2} + \eta \tan^{-1}(\xi'/\eta) - \xi' \right]_{\xi'=-\xi_0-\xi}^{\xi'=\xi_0-\xi}, \end{aligned} \tag{4.24}$$

with the latter term giving the contribution from the integral along the base, that is the piece $-\xi_0 \leq \xi \leq \xi_0$ of the ξ axis. For this to represent a steady vortex dipole embedded in a flow $(-c_0, 0)$ at infinity we need the total streamfunction $\Psi_0 = \psi_0 + c_0 \eta$ to be zero on the contour C_0 . This condition enables C_0 to be found for a given c_0 , for example using a collocation method as described in Saffman & Tanveer (1982). (In

equation (4.24) we correct a misprint in Saffman & Tanveer (1982), noting that their streamfunction is taken with the opposite sign to ours.)

With C_0 known at least numerically, we can use a coordinate system based on arc-length σ along the contour and a coordinate χ that measures distance perpendicular to the contour C_0 ; see figure 4. The corresponding metric is then

$$ds^2 = d\chi^2 + h^2 d\sigma^2, \quad h = 1 + \kappa(\sigma)\chi, \tag{4.25}$$

where κ is the curvature of the curve C_0 at the point given by σ . We need to recast the first equation of (4.22) in a contour dynamics setting. We take the non-zero constant vorticity to be $\varpi = 1$ in the upper half-plane, confined by a time-dependent contour which we call C and suppose (with mild abuse of notation) given by a function $\chi = C(\sigma, \tau)$ (see figure 4). The situation in the lower half-plane is mirror symmetric. Now C is a material curve and so

$$\frac{D}{D\tau} (C(\sigma, \tau) - \chi) = 0, \tag{4.26}$$

or

$$\frac{\partial C}{\partial \tau} = \mathbf{U} \cdot \left(\mathbf{n} - \frac{1}{h} \frac{\partial C}{\partial \sigma} \mathbf{t} \right) \Big|_{\chi=C(\sigma, \tau)}, \tag{4.27}$$

with $\mathbf{t} = h\nabla\sigma$ and $\mathbf{n} = \nabla\chi$ being tangential and normal unit vectors. The unperturbed problem has $C(\sigma, \tau) = C_0(\sigma, \tau) \equiv 0$ and $\mathbf{U}_0 \cdot \mathbf{n} = 0$. At the first order we set $C(\sigma, \tau) = C_1(\sigma, \tau) + \dots$, $\mathbf{U} = \mathbf{U}_0 + \mathbf{U}_1 + \dots$ to obtain in the linear approximation,

$$\frac{\partial C_1}{\partial \tau} = \left(C_1 \frac{\partial \mathbf{U}_0}{\partial \chi} + \mathbf{U}_1 \right) \cdot \mathbf{n} - \frac{\partial C_1}{\partial \sigma} \mathbf{U}_0 \cdot \mathbf{t} \tag{4.28}$$

evaluated on the curve C_0 given by $\chi = 0$. With the use of the streamfunction we write

$$\mathbf{U}_0 = -\frac{1}{h} \frac{\partial \Psi_0}{\partial \sigma} \mathbf{n} + \frac{\partial \Psi_0}{\partial \chi} \mathbf{t} \tag{4.29}$$

and after a short calculation obtain

$$\frac{\partial C_1}{\partial \tau} + \frac{\partial}{\partial \sigma} \left(C_1 \frac{\partial \Psi_0}{\partial \chi} \right) = \mathbf{U}_1 \cdot \mathbf{n} \tag{4.30}$$

again evaluated on C_0 .

Setting

$$\Phi_1 = C_1 \frac{\partial \Psi_0}{\partial \chi} \equiv C_1 \mathbf{U}_0 \cdot \mathbf{t} \tag{4.31}$$

we can write the equation in perhaps the most intuitive form

$$\frac{\partial \Phi_1}{\partial \tau} + \mathbf{U}_0 \cdot \mathbf{t} \frac{\partial \Phi_1}{\partial \sigma} = (\mathbf{U}_0 \cdot \mathbf{t})(\mathbf{U}_1 \cdot \mathbf{n}). \tag{4.32}$$

This represents advection of vorticity flux $\Phi_1(\sigma, \tau)$ between curves C_0 and C_1 along the unperturbed curve C_0 , with a source term that involves the perpendicular component of the perturbation velocity \mathbf{U}_1 . Note that as we approach the trailing stagnation point, where vorticity will peel off into the flow, $\mathbf{U}_0 \cdot \mathbf{t} \rightarrow 0$ and so the

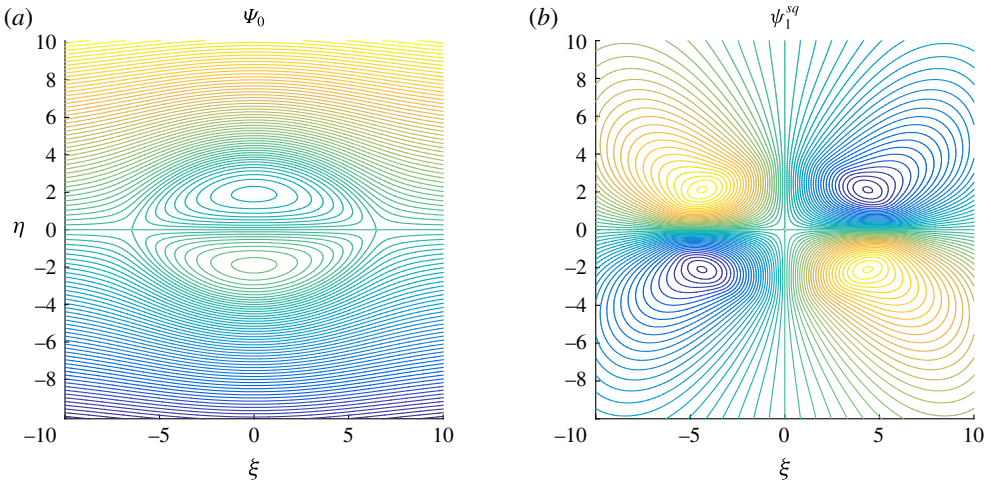


FIGURE 5. (Colour online) In (a) the flow field for the Sadovskii vortex is depicted, given by curves of constant $\Psi_0 = \psi_0 + c_0\eta$ (4.24) in the (ξ, η) plane, and in (b) the flow ψ_1^{sq} (4.18) driven by vortex curvature, which is roughly of stagnation point form.

source term on the right-hand side is suppressed and the quantity Φ_1 will be seen to converge, even though C_1 must diverge there.

With the key machinery in place, we indicate the numerical solution that aims to fix p_1 , through time stepping the partial differential equation (PDE) (4.32) until it can be made to converge to a steady state. Before we time step we evaluate the boundary of the Sadovskii vortex from (4.24) following Saffman & Tanveer and express this as a curve $\rho = \rho_0(\theta)$ in polar coordinates in the (ξ, η) -plane; the resulting flow field is depicted in figure 5(a). From this we may evaluate \mathbf{t} and \mathbf{n} along C_0 relative to polar coordinates. Then, for the left-hand side of (4.32) we need $\mathbf{U}_0 \cdot \mathbf{t}$ which is obtained from the Sadovskii streamfunction in (4.24) with $\Psi_0 = \psi_0 + c_0\eta$ by finite differencing of ψ_0 as obtained numerically. Turning to the right-hand side of (4.32), \mathbf{U}_1 contains several components from (4.19), in order,

$$\mathbf{U}_1 = \mathbf{U}_1^{frame} + \mathbf{U}_1^{\mathcal{G}} + \mathbf{U}_1^{sq} + c_1 \mathbf{U}_1^{trans} - c_0 p_1 \mathbf{U}_1^{scale}. \quad (4.33)$$

The most straightforward of these are expressed in polar coordinates as

$$\mathbf{U}_1^{frame} = -c_0 \rho \cos \theta (\cos \theta \hat{\boldsymbol{\rho}} - \sin \theta \hat{\boldsymbol{\theta}}), \quad (4.34)$$

$$\mathbf{U}_1^{sq} = -\frac{1}{\rho} \frac{\partial \psi_1^{sq}}{\partial \theta} \hat{\boldsymbol{\rho}} + \frac{\partial \psi_1^{sq}}{\partial \rho} \hat{\boldsymbol{\theta}}, \quad (4.35)$$

$$\mathbf{U}_1^{trans} = -\cos \theta \hat{\boldsymbol{\rho}} + \sin \theta \hat{\boldsymbol{\theta}}, \quad \mathbf{U}_1^{scale} = -\rho \hat{\boldsymbol{\rho}}. \quad (4.36a,b)$$

The term arising from vortex line curvature is ψ_1^{sq} which is a fixed flow field that can be evaluated once at the start of the computation. This is done rather crudely by evaluating $\partial^2 \psi_0 / \partial \eta^2$ using finite differences, then applying \mathcal{G} by approximating the integral as a finite sum over grid points and finally finite differencing again. Streamlines of the resulting flow field are shown in figure 5(b); this has an approximate stagnation point form, pressing the two lobes of the vortex together.

Finally as we time step the PDE (4.32) the only term that cannot be pre-calculated is the feedback $\mathbf{U}_1^{\mathcal{G}}$ which is the flow arising from $\mathcal{G}\varpi_1$, from the perturbed contour and a functional of $C_1(\sigma, \tau)$. Now the unperturbed contour is $\rho = \rho_0(\theta)$ in polar coordinates, and the gap between this and the perturbed contour, $\chi = C_1(\sigma, \tau)$, gives essentially a vortex sheet which has to be integrated as in (4.14) to obtain the corresponding flow. At a point $(\rho_0(\theta), \theta)$ on the contour the normal component that we need may be written as an integral over the contour, in terms of the dummy variable θ' ,

$$\begin{aligned} \mathbf{U}_1^{\mathcal{G}} \cdot \mathbf{n} &= \frac{1}{2\pi} \\ &\times \int_0^\pi \left\{ \frac{\rho_0 \rho'_0 \sin(\theta - \theta') + (\partial_\theta \rho_0)[\rho_0 - \rho'_0 \cos(\theta - \theta')]}{(\rho_0 - \rho'_0)^2 + 4\rho_0 \rho'_0 \sin^2 \frac{1}{2}(\theta - \theta')} \right\} \frac{j}{j'} C'_1 - \frac{1}{2} \cot \frac{1}{2}(\theta - \theta') C_1 \Big\} d\theta' \\ &- \frac{1}{2\pi} \\ &\times \int_0^\pi \left\{ \frac{\rho_0 \rho'_0 \sin(\theta + \theta') + (\partial_\theta \rho_0)[\rho_0 - \rho'_0 \cos(\theta + \theta')]}{(\rho_0 - \rho'_0)^2 + 4\rho_0 \rho'_0 \sin^2 \frac{1}{2}(\theta + \theta')} \right\} \frac{j}{j'} C'_1 - \frac{1}{2} \cot \frac{1}{2}(\theta + \theta') C_1 \Big\} d\theta' \\ &+ \pi^{-1} \log \left(\tan \frac{1}{2}\theta \right) C_1, \end{aligned} \tag{4.37}$$

where $\partial_\theta \rho_0 = d\rho_0/d\theta$, a prime denotes evaluation with respect to the dummy variable θ' and the Jacobian is given by

$$j(\theta)^{-1} \equiv \frac{d\sigma}{d\theta} = \left[\left(\frac{d\rho_0}{d\theta} \right)^2 + \rho_0^2 \right]^{1/2}. \tag{4.38}$$

With this in place, we time step the PDE (4.32) in terms of $\Phi_1(\sigma, \tau)$ by evaluating $\mathbf{U}_1^{\mathcal{G}} \cdot \mathbf{n}$ at each time τ and looking up all the other components of the flow field. We need to allow c_1 and p_1 to converge so that $\Phi_1(\sigma, \tau)$ becomes steady as $\tau \rightarrow \infty$, thus avoiding secular behaviour. We have freedom about how this is done: any two conditions that fix the scale and the ξ -location of the vortex will suffice. We choose

$$\mathbf{U}_1 \cdot \mathbf{n} = 0 \quad \text{at } \theta = 0, \pi/2 \tag{4.39}$$

and so once all the components of $\mathbf{U}_1 \cdot \mathbf{n}$ are found the calculation of c_1 and p_1 is straightforward. The result of time stepping is that Φ_1 converges to a τ -independent profile, depicted in figure 6(a) with $p_1 \simeq 0.74$. This is in line with the theoretical value $p_1 = 3/4$ needed for energy conservation. Note that of the two driving terms, with only \mathbf{U}_1^{sq} we obtain $p_1 = 0.37$, and with only \mathbf{U}_1^{frame} , $p_1 = 0.35$, so these each account for approximately half the effect.

Finally, we remark on the formula (4.37): the feedback on the flow because the contour C differs a little from C_0 amounts to calculating the flow from a vortex sheet of strength $C_1(\sigma, \tau)$ along the curve C_0 . Such an integral has to be taken as a principal value, and here we have done this by removing explicitly the singular components from the integrands in (4.37), which are then placed in the final $\log(\tan(1/2)\theta)$ term. Taking the principal value is appropriate as at a point σ on the vortex sheet/thin layer the transverse flows generated by the vorticity for $\sigma' > \sigma$ and $\sigma' < \sigma$ locally cancel

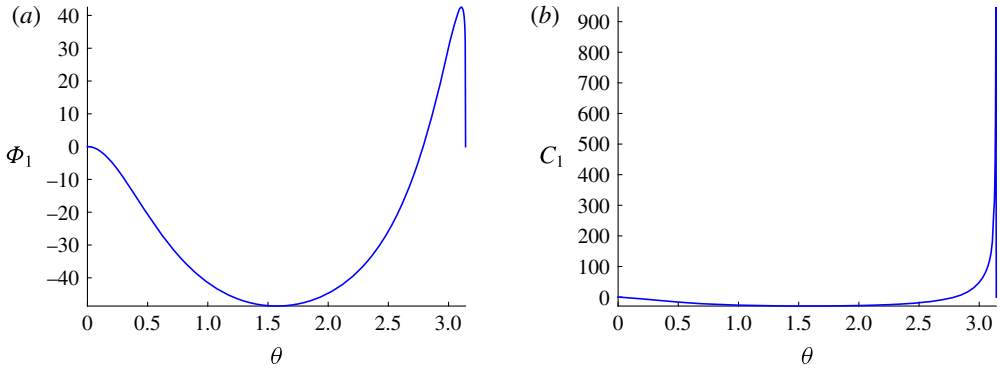


FIGURE 6. (Colour online) The steady correction to the Sadvskii vortex is shown by plots of (a) Φ_1 and (b) C_1 (see (4.31)) as functions of the polar angle θ in the (ξ, η) plane, valid in the asymptotic limit $\tau \rightarrow \infty$.

out. However at $\theta = 0$ and $\theta = \pi$, this argument fails: the vortex sheet comes to an abrupt end and in fact changes sign. (The curvature singularities and singular flow field here in the underlying Sadvskii vortex are explained in depth in Saffman & Tanveer (1982)). This explains the presence of the logarithmic singularity at $\theta = 0, \pi$ in the term $\log(\tan(1/2)\theta)$. In our calculations we have taken $\mathbf{U}_1 \cdot \mathbf{n} = 0$ at $\theta = 0$ in (4.39) which keeps $C_1 = 0$ there (see (4.30)) and removes immediate difficulties with this term. For $\theta = \pi$ the singular term is present, and is part of the flow field that leads to the ejection of vorticity from the rear of the vortex pair.

5. Numerical simulation of the snail

We complement the analysis offered in previous sections with the results of a direct numerical simulation of the evolving axisymmetric vortex dipole. The results of numerical simulation presented in this section both confirm the leading-order analysis undertaken in previous sections, and provide crucial insight into the non-asymptotic regime by showing how the snail reliably emerges and stably evolves from typical initial conditions of the appropriate symmetry.

5.1. Set-up

We simulate AFWOS subject to the following conditions: vorticity is anti-symmetric about the $z = 0$ plane and vorticity is non-zero only in a small region (which possibly moves over time, and may lie far from the $r = 0$ axis). The configuration of the simulation is represented at each time step by the values of θ -vorticity within a small square region of the plane. Time stepping is implemented via the fourth-order Runge–Kutta scheme, with the components of the Euler equation recovered from the vorticity via the Biot–Savart law. Calculations are undertaken in a local Fourier basis to preserve as much spatial accuracy as possible, both in the simulation, and in the computation of quantities of interest afterwards. While this Fourier basis has many convenient features favouring the speed and accuracy of the simulation, there are a number of complications introduced by the mismatch between the periodic nature of the basis and the infinite domain of the cylindrical coordinate system.

Every initial configuration we simulated evolved into a ‘snail’; we here examine the trajectory of one configuration in depth. We describe the initial conditions here and

explain why they could be expected to evolve into a snail in a particularly direct and smooth way. Specifically, the initial condition consists of two anti-parallel vortex rings, each of which has vorticity which is the product of the radial coordinate r , with a smooth transition function which is close to 1 inside a torus and close to 0 outside it: an appropriately shifted and scaled error function applied to the distance from circular centreline of each torus. Specifically, given cylindrical coordinates (r, z, θ) , the initial condition for the vorticity in the θ direction is defined by

$$\omega = r \operatorname{erf} \left(\frac{\sqrt{(r - 0.7)^2 + (z - 0.32)^2}}{0.06} - 3 \right) - r \operatorname{erf} \left(\frac{\sqrt{(r - 0.7)^2 + (z + 0.32)^2}}{0.06} - 3 \right). \quad (5.1)$$

Thus our initial conditions have vortex rings at $(r, z) = (0.7, \pm 0.32)$, each of radius $3 \times 0.06 = 0.18$, and with the transition from the interior to the exterior of each ring occurring over roughly 0.06 distance. The vorticity is chosen to be nearly homogeneous (before the r scaling) inside each vortex ring so that when the snail sheds the outer layers of each ring, the vorticity will become increasingly constant inside the evolving snail.

Starting with vortex rings at radius 0.7, the simulation was run until the radius at the centre of the snail was 9.6. The diameter of the vortex tube was initially 0.36 and was finally 0.12 in the radial direction and 0.039 in the z direction, having shed 55% of its circulation. Because our simulation repeatedly increases resolution so that the snail remains several hundred grid points across, the grid size decreases from 0.002 down to 0.00035 over the course of a run, and would continue to decrease as r increases. The grid used is 800×800 pixels and this is centred on the vortex rings; the tail is passive, and allowed to trail behind the vortex dipole, and out of the box in the simulation. As we do many operations in a local ‘periodic’ box, i.e. using a Fourier basis, in many operations we use masking around the edge to avoid the numerical periodicity interfering with the actual cylindrical geometry. Stability concerns dictate that the evolving vortex rings can move at most a fraction of a grid space in each simulation time step, which employs a fourth-order Runge–Kutta scheme, meaning that the cost of continuing for larger r would continue to increase rapidly, and running over a much wider range of radii is infeasible. Nonetheless, we present results in § 5.2 showing that already within the scope of this simulation, the configuration converges rapidly to the expected behaviour of the snail.

The most time consuming part of the code is, given the vorticity distribution in the box in the local (r, z) -plane, to compute the corresponding flow field. This involves a convolution in the z -direction which may be done in Fourier space, the problem being separable in the axial direction. (A box of twice the vertical extent and masking are used to avoid any spurious effects of the numerical periodicity interfering with the convolution.) However in the radial direction it is necessary to undertake the convolution explicitly and to compute the appropriate elliptic integrals in this Green’s function (with the use of both Matlab’s built-in function and an expansion valid for points close to the source). Because of the expense in evaluating the elliptic integrals, as the window following the vortex moves and also zooms in (through six different resolutions in the run presented here), much of this Green’s function data are reused by resampling on the new grid.

5.2. Results

We describe the results of the simulation here, both qualitatively and by quantitatively verifying scaling laws. In figure 7 we show several snapshots: in each case, we are

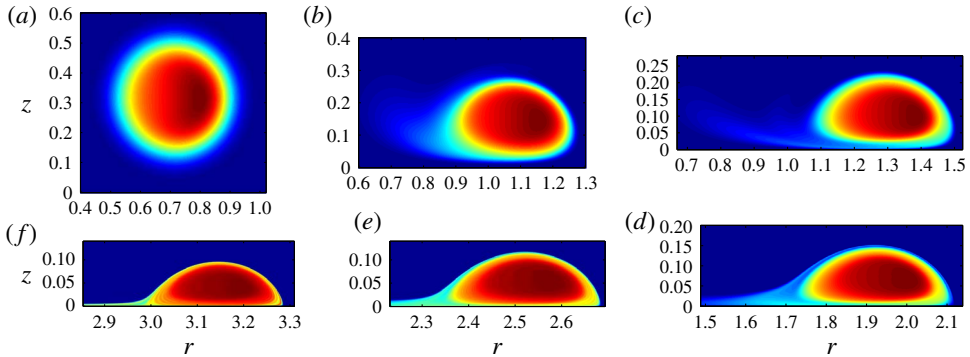


FIGURE 7. Development of the snail, shown at times $t = 0, 15, 21.2, 34.2, 44.3, 54$ clockwise from the upper left corner. Vorticity ω is plotted in the (r, z) plane at each time, scaled on the maximum value (red) in each plot, with zero blue.

displaying a slice in the (r, z) plane, depicting vorticity in the θ direction (note the changing scale of each panel). Initially, we have two vortex tubes, relatively diffuse, and separated from each other across the $z=0$ symmetry plane. They quickly move towards each other, without moving much away from the z -axis yet. When well separated the tubes are driven together by a converging flow along the binormal, as in the filament computations of Pumir & Siggia (1987). Soon, the tubes have essentially hit the $z=0$ symmetry plane and begin to shed vorticity into the tail as the dipole expands and the tubes are stretched. From here, the recognizable snail shape develops.

The lateral extent of the vortex tubes decreases significantly as the tubes are stretched away from the axis and shed volume; the thickness of the shed tail also decreases relative to the thickness of the snail, since otherwise the snail would lose all its volume in finite time. While the speed of the snail increases with distance from the axis, the shed tail is essentially stationary, having velocity $\sim r^{-3/2}$, see equation (2.2). The element of tail shed at a certain radius from the z axis will maintain its thickness at that radius forever. This justifies the neglect in the simulation of those parts of the tail that have lagged behind and thereafter fall outside the simulation box.

We next measure several aspects of the simulated snail and confirm that they follow the expected scalings. One of the key surprises of the snail is that its velocity increases without limit. Explicitly, the prediction is that the radius of each vortex ring should grow superlinearly with time, now adopting the dipole position $R(t)$ used earlier, as

$$R \sim (t + c)^{4/3}, \quad (5.2)$$

where the additive constant c captures the fact that the start time of the simulation is arbitrary. To demonstrate this $4/3$ -power relation, we instead plot in figure 8 the $3/4$ -power of both sides, $R^{3/4} \sim t + c$: in blue is the radius of the centre of each vortex tube as a function of time; drawn below in black is an arbitrary line to help verify visually the claim that, asymptotically, $R^{3/4} \sim t + c$. Thus the snail does in fact accelerate over time, with radius proportional to $t^{4/3}$ and velocity proportional to its derivative, $t^{1/3}$.

The other main prediction of the snail model concerns how volume is shed, and how the dimensions of the tube decrease faster than mere stretching would allow. Since we

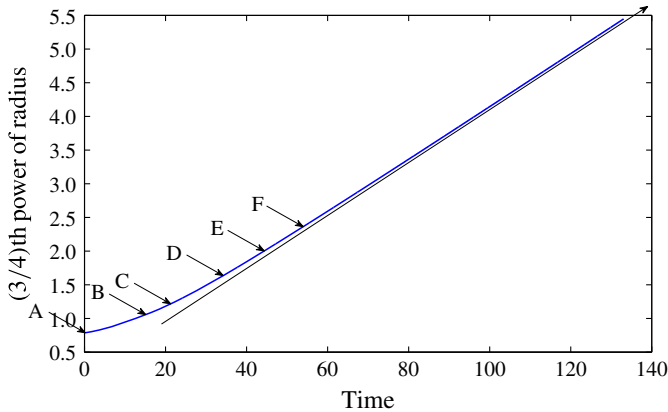


FIGURE 8. (Colour online) $R^{3/4}$ versus time, compared with a linear asymptote. The letters correspond to the frames of figure 7.

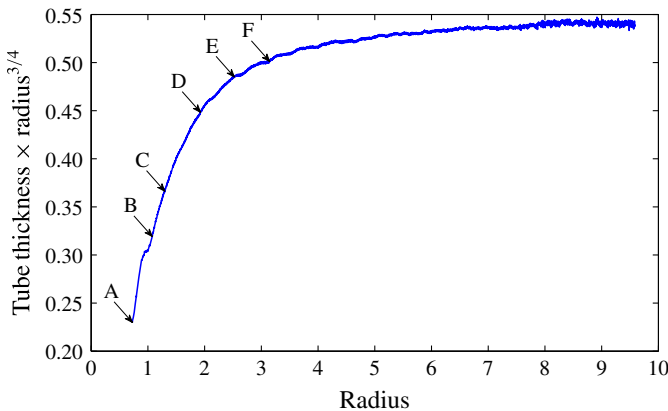


FIGURE 9. (Colour online) Maximum dipole thickness in the z direction times $R^{3/4}$ as a function of R . The letters correspond to the frames of figure 7.

expect the dimensions of the snail to decay as $R^{-3/4}$, we plot in figure 9 the thickness in the z direction, defined as the maximum z extent of the region where vorticity exceeds 70% of its maximum, times $R^{3/4}$; our result is that this quantity does indeed approach a constant as the simulation progresses.

Finally, one more qualitative prediction which is supported numerically is that the ‘edge’ of the snail, that is, the width of the transition from high vorticity to low vorticity, sharpens quickly, perhaps exponentially, with time, associated with the emergence of the Sadovskii structure. The thickness of both the outer edge, adjacent to the exterior flow, and the edge adjacent to the plane of symmetry, quickly fall below the grid size, 0.002 initially and decreasing over the course of the simulation.

5.2.1. Other initial conditions

It is revealing to consider the sensitivity of our dipole to small changes in the initial conditions. Two alternatives are especially worth discussing: first if the snail is inhomogeneous, how do ‘lumps’ in the snail translate to lumps in the tail, or affect

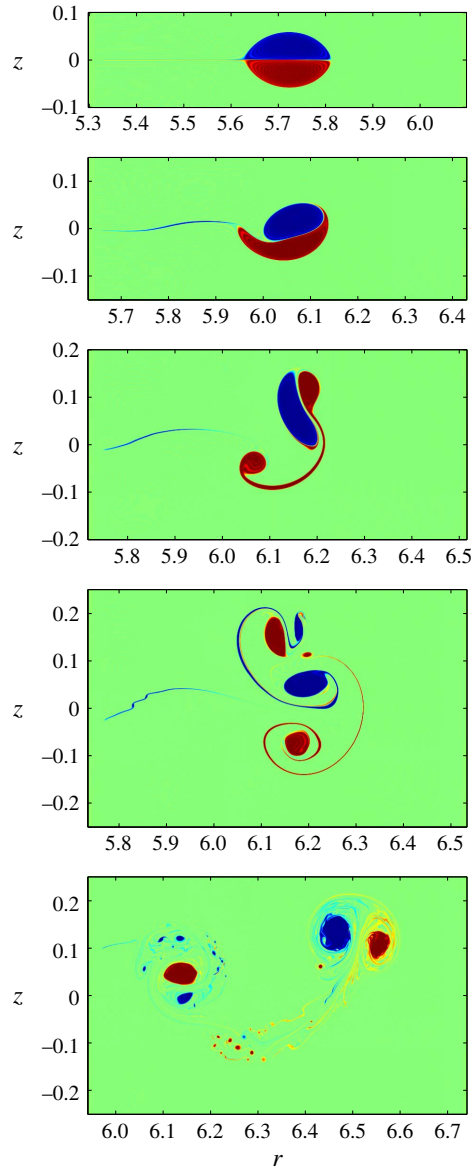


FIGURE 10. Breakup of the snail by instability when mirror symmetry of the dipole is not imposed.

the overall scaling? Secondly, how does the snail react to symmetry breaking, and in particular, violating the anti-symmetry about the plane $z = 0$?

The brief answers are that: variations of initial conditions that preserve symmetry do not much affect the snail, which appears to be a very robust phenomenon; however symmetry breaking rapidly amplifies, leading to a breakdown of the dipole, where radial stretching not only stops accelerating, but typically stops entirely – thus the symmetries of the snail seem fundamental to its evolution. We show this break-up in figure 10. It is significant that precise symmetry is needed to maintain vorticity

growth, a point that is particularly important in the search for vortical structures which blow up in finite time. It is interesting that in this example the break-up of the dipole produces two smaller dipole-like structures.

6. Discussion

We have in this paper presented a model for vorticity growth in anti-parallel vortex structures in axisymmetric flow without swirl. The model provides an Euler flow which achieves the maximum possible growth of $|\omega|_{max}$ as $t^{4/3}$. The new feature of this work is the explicit role of vortex erosion, leading to scalings quite distinct from those associated with intact vortex tubes. The governing assumption behind the new scaling is the local conservation of energy following Lagrangian parcels of fluid. Our analysis has been restricted to a symmetric dipole arrangement of equal and opposite vortical eddies, which leads to the Sadvovskii structure. We believe that asymmetric 2-D dipoles of Sadvovskii type, with constant vorticity in each eddy but differing circulations, are likely to exist (Meleshko & van Heijst 1994). Such a dipole would move on a circular path, and hence would lead to a more complicated centre curve involving non-zero torsion. This greatly complicates the analysis but could be an interesting generalization.

The calculations for the axisymmetric case indicate that the breaking of mirror symmetry leads to break-up of the dipole. Also non-axisymmetric instabilities occur when equal and opposite vortex rings collide. Thus the ultimate growth calculated here may not be observed in actual flows irrespective of the size of the Reynolds number.

There has been no discussion here of the analogous problem for Navier–Stokes flows, and in particular of maximizing enstrophy growth for a given enstrophy. This problem was considered in Lu & Doering (2008) and it is interesting that colliding vortex rings arise there as optimizing flows. However the length scale of these optimizers tends to zero with viscosity and so a bound on growth for Euler does not exist.

What are the implications of these calculations for more general Euler flows? The ‘swirl’ which is absent in the present model amounts to flow in rings along the axis of the dipole. Axisymmetric flow with swirl can, according to Luo & Hou (2014), blow up in finite time in the presence of an impenetrable boundary. In \mathbb{R}^3 the situation is unclear. The present model will, with the addition of swirl, change significantly owing to the generation of axial vorticity by the z -derivative of the centrifugal pressure. More general non-axisymmetric 3-D dipole models must cope with the generation of axial flow by the axial pressure gradient produced as the dipole stretches differentially. Variation of this flow along the axis will then modify the axial vorticity. General anti-parallel geometries must again cope with the loss of a symmetry plane. The time scale of this breaking of symmetry will compete with the effects of vortex stretching.

The structures we have examined in this paper may play a role in future studies of more rapid vorticity growth in \mathbb{R}^3 , as we have suggested in Childress (2008). The more modest growth we obtain here is a direct consequence of the axisymmetric geometry. In a subsequent paper we will apply many of the ideas of the present paper to a non-axisymmetric geometry, and discuss the role of axial flow on the resulting growth of vorticity.

Acknowledgement

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Appendix A. Potential flow past an expanding torus of constant volume

A torus of radius R and cross-sectional area πa^2 expands radially (i.e. outward in the plane of symmetry) in a perfect inviscid fluid, $R = R(t)$, while maintaining a constant volume. What is the resulting irrotational flow field?

We first consider the potential

$$\phi = -\frac{R}{4\pi} \int_0^{2\pi} \frac{d\theta}{\sqrt{R^2 + r^2 - 2Rr \cos 2\theta + z^2}}, \quad (\text{A } 1)$$

representing a uniform distribution of sources over the circle $z = 0$, $r = R$. This can be brought into the form

$$\phi = -\frac{R}{\pi} \int_0^{\pi/2} \frac{d\theta}{\sqrt{(R+r)^2 + z^2 - 4Rr \sin^2 \theta}}, \quad (\text{A } 2)$$

or

$$\phi = -\frac{R}{\pi P} \int_0^{\pi/2} \frac{d\theta}{\sqrt{1 - k^2 \sin^2 \theta}} = -\frac{R}{\pi P} K(k), \quad (\text{A } 3)$$

where

$$P = \sqrt{(R+r)^2 + z^2}, \quad k^2 = \frac{4Rr}{(R+r)^2 + z^2}. \quad (\text{A } 4a,b)$$

Near $k = 1$ we have (Carlson & Gustafson 1985)

$$K(k) = \sum_{n=0}^{N-1} \left[\frac{(\frac{1}{2})_n}{n!} \right]^2 \left[\log \frac{1}{k'} + \psi(1+n) - \psi(1/2+n) \right] (k')^{2n} + O(k')^{2N} \log k', \quad (\text{A } 5)$$

where

$$\left(\frac{1}{2}\right)_n = \frac{\Gamma(n + \frac{1}{2})}{\Gamma(\frac{1}{2})}, \quad \psi(1) - \psi(1/2) = 2 \log 2, \quad (\text{A } 6a,b)$$

$$\psi(1+n) - \psi(1/2+n) = 2 \left[\log 2 - 1 + \frac{1}{2} - \dots - \frac{1}{2n-1} + \frac{1}{2n} \right], \quad n \geq 1, \quad (\text{A } 7)$$

and

$$k' = \sqrt{1 - k^2}. \quad (\text{A } 8)$$

Going over to local coordinates we have $r = R + x$, $z = y$, $\rho^2 = x^2 + y^2$. Then we have

$$P = 2R \sqrt{1 + x/R + \frac{1}{4} \rho^2 / R^2}, \quad k' = \rho / P. \quad (\text{A } 9a,b)$$

Expanding through terms of order $(\rho/R)^2$ we have

$$\begin{aligned} \phi = & -\frac{1}{2\pi} \left[1 - \frac{1}{2} \frac{x}{R} + \frac{3}{8} \left(\frac{x}{R}\right)^2 - \frac{1}{8} \left(\frac{\rho}{R}\right)^2 \right] \\ & \times \left[\log \frac{8R}{\rho} + \frac{1}{2} \frac{x}{R} + \frac{9}{4} \left(\log \frac{8R}{\rho} - 1 \right) \frac{\rho^2}{4R^2} \right] + o\left(\frac{\rho^2}{R^2}\right). \end{aligned} \quad (\text{A } 10)$$

This gives the ordering

$$2\pi\phi = -\log \frac{8R}{\rho} + \left[\frac{x}{2} \left(\log \frac{8R}{\rho} - 1 \right) \right] \frac{1}{R} + \dots \tag{A 11}$$

We will use these terms in the expansion of ϕ to solve the problem of the torus of constant volume.

We seek the potential flow past a torus expanding so that $R(t)$ increases with time, with the radius $a(t)$ of the cross-section satisfying

$$\frac{\dot{a}}{a} = -\frac{1}{2} \frac{\dot{R}}{R}. \tag{A 12}$$

We use the fact that if ϕ solves Laplace's equation in three dimensions, then so does $R\phi' = x\phi_x + y\phi_y + z\phi_z$ or, with radial symmetry,

$$R\phi' = r\phi_r + z\phi_z = R\phi_x + \rho\phi_\rho, \tag{A 13}$$

or

$$\phi' = \phi_x + R^{-1}\rho\phi_\rho. \tag{A 14}$$

Using (A 11) for the expansion of ϕ in (A 14) we see that

$$2\pi\phi' \equiv \Phi + \frac{1}{R} = \frac{x}{\rho^2} + \frac{1}{2R} \log \frac{8R}{\rho} - \frac{1}{2R} \frac{x^2}{\rho^2} + \frac{1}{R} + O(R^{-2}). \tag{A 15}$$

Thus Φ is a building block of the local potential for a cylindrical cross-section. Indeed

$$-\dot{R}(x + a^2\Phi) \sim -\dot{R}x(1 + a^2/\rho^2) \tag{A 16}$$

is the potential for uniform flow over a cylinder.

Now in the neighbourhood of infinity we see that

$$\phi' \sim \frac{1}{2\sqrt{r^2 + z^2}}, \tag{A 17}$$

giving a net source flux of -2π . We can check that this is consistent with flux out of the surface of the torus. Indeed $x/(2\pi\rho^2)$ contributes

$$\frac{1}{2\pi} \int_0^{2\pi} (-\cos\theta/\rho^2) 2\pi\rho(R + \rho\cos\theta) d\theta = -\pi, \tag{A 18}$$

and $(4\pi R)^{-1} \log(8R/\rho)$ contributes

$$\frac{2\pi R}{4\pi R} \int_0^{2\pi} (-1/\rho)\rho d\theta = -\pi. \tag{A 19}$$

To obtain a potential free of net source strength we must then add on ϕ/R , and so the potential of the expanding torus, at the point where its cross-sectional radius is a , relative to the fluid at infinity (not co-moving), is

$$\phi_{torus} = -2\pi\dot{R}a^2(\Phi + \phi/R) \sim a^2\dot{R} \left[-\frac{x}{\rho^2} + \frac{1}{2R} \log \frac{8R}{\rho} + \frac{1}{2R} \frac{x^2}{\rho^2} \right] + O(a^2\dot{R}/R^2). \tag{A 20}$$

Recalling (A 12), it is readily seen that the exhibited terms lead to the appropriate normal velocity at the instantaneous surface of the torus.

REFERENCES

- BATCHELOR, G. K. 1956 On steady laminar flow with closed streamlines at large Reynolds number. *J. Fluid Mech.* **1**, 177–190.
- BUSTAMANTE, M. D. & KERR, R. M. 2008 3D Euler about a 2D symmetry plane. *Physica D* **237**, 1912–1920.
- CARLSON, B. C. & GUSTAFSON, J. L. 1985 Asymptotic expansion of the first elliptic integral. *SIAM J. Math. Anal.* **16**, 1072–1092.
- CHERNYSHENKO, S. I. 1988 The asymptotic form of the stationary separated circumfluence of a body at high Reynolds-numbers. *PMM J. Appl. Math. Engng* **52**, 746–753.
- CHILDRESS, S. 1966 Solutions of Euler's equations containing finite eddies. *Phys. Fluids* **9**, 860–872.
- CHILDRESS, S. 1987 Nearly two-dimensional solutions of Euler's equations. *Phys. Fluids* **30**, 944–953.
- CHILDRESS, S. 2008 Growth of anti-parallel vorticity in Euler flows. *Physica D* **237**, 1921–1925.
- CHILDRESS, S. 2009 S. Constraints on stretching by paired vortex structures I. Kinematics, II. The asymptotic dynamics of blow-up in three dimensions. Preliminary reports available at <http://www.math.nyu.edu/faculty/childres/preprints.html>.
- GIBBON, J. D. 2008 The three-dimensional Euler equations: Where do we stand? *Physica D* **237**, 1894–1904.
- GRAFKE, T. & GRAUER, R. 2013 Lagrangian and geometric analysis of finite-time Euler singularities. *Procedia IUTAM* **7**, 32–56.
- HORMOZ, S. & BRENNER, M. P. 2012 Absence of singular stretching of interacting vortex filaments. *J. Fluid Mech.* **707**, 191–204.
- HOU, T. Y. & LI, R. 2008 Blowup or no blowup? The interplay between theory and numerics. *Physica D* **237**, 1932–1944.
- KERR, R. M. 2013 Bounds on Euler from vorticity moments and line divergence. *J. Fluid Mech.* **729**, R2.
- LIM, T. T. & NICKELS, T. B. 1992 Instability and reconnection in the head-on collision of two vortex rings. *Nature* **357**, 225–227.
- LU, L. & DOERING, C. R. 2008 Limits on enstrophy growth for solutions of the three-dimensional Navier–Stokes equations. *Indiana Univ. Math. J.* **57**, 2693–2727.
- LUO, G. & HOU, T. Y. 2014 Potentially singular solutions of the 3D axisymmetric Euler equations. *Proc. Natl Acad. Sci. USA* **111**, 12968–12973.
- MAJDA, A. J. & BERTOZZI, A. L. 2001 *Vorticity and Incompressible Flow*. Cambridge University Press.
- MELESHKO, V. V. & VAN HEIJST, G. J. F. 1994 On Chaplygin's investigations of two-dimensional vortex structures in an inviscid fluid. *J. Fluid Mech.* **272**, 157–182.
- MOORE, D. W., SAFFMAN, P. G. & TANVEER, S. 1988 The calculation of some Batchelor flows: the Sadvovskii vortex and rotational corner flow. *Phys. Fluids* **31**, 978–990.
- OSHIMA, Y. 1978 Head-on collision of two vortex rings. *J. Phys. Soc. Japan* **44**, 329–331.
- PIERREHUMBERT, R. T. 1980 A family of steady, translating vortex pairs with distributed vorticity. *J. Fluid Mech.* **99**, 129–144.
- PRANDTL, L. 1952 *Essentials of Fluid Mechanics*. Blackie.
- PUMIR, A. & KERR, R. M. 1987 Numerical simulation of interacting vortex tubes. *Phys. Rev. Lett.* **58**, 1636–1639.
- PUMIR, A. & SIGGIA, E. 1987 Vortex dynamics and the existence of solutions to the Navier–Stokes equations. *Phys. Fluids* **30**, 1606–1626.
- RILEY, N. 1998 The fascination of vortex rings. *Appl. Sci. Res.* **58**, 169–189.
- SADOVSKII, V. S. 1971 Vortex regions in a potential stream with a jump of Bernoulli's constant at the boundary. *Prikl. Mat. Mekh.* **35**, 773–779.
- SAFFMAN, P. G. & TANVEER, S. 1982 The touching pair of equal and opposite vortices. *Phys. Fluids* **25**, 1929–1930.
- SHARIF, K., LEONARD, A. & FERZIGER, J. H. 2008 A contour dynamics algorithm for axisymmetric flows. *J. Comput. Phys.* **227**, 9044–9062.
- SHELLEY, M. J., MEIRON, D. I. & ORSZAG, S. A. 1993 Dynamical aspects of vortex reconnection of perturbed anti-parallel vortex tubes. *J. Fluid Mech.* **246**, 613–652.